

# AVIATION

OCTOBER 1, 1923

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ADVANCE NEWS  
OF THE RACES  
ST. LOUIS ANNOUNCEMENT  
ISSUE

VOLUME  
XV

## SPECIAL FEATURES

NUMBER  
14

ARMY AND NAVY ENTRIES FOR PULITZER RACE  
BIOGRAPHIC NOTES OF ARMY AND NAVY ENTRANTS  
NEW WRIGHT T3 AVIATION ENGINE  
HUFF DALAND ADVANCED TRAINING PLANE

THE GARDNER, MOFFAT CO., INC.  
HIGHLAND, N. Y.  
225 FOURTH AVENUE, NEW YORK

Entered as Second-Class Matter, Nov. 22, 1920, at the Post Office at Highland, N. Y.  
under Act of March 3, 1879.

# LIGHTNESS

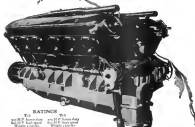
THE weight per H.P. of the Wright T-3 engine is better than guaranteed for any other water cooled engine being built. This low weight to power ratio improves speed, climb, ceiling and maneuverability. Greater useful load can be carried for a given wing loading.

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150 H.P. 1000 ft. per min.  
150 H.P. 1000 ft. per min.  
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# WRIGHT MODELS T ENGINES

## Record-Breakers— ALL

### 1923

Pulitzer Trophy Race -- won by.....?

Curtiss Marine Trophy Race -- won by.....?

### 1922

Pulitzer Trophy Race -- won by Lieutenant Maughan, pilot, average 206 miles per hour.

Curtiss Marine Trophy Race -- won by Lieutenant Gorton, pilot, average 112 miles per hour.



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on all Record-breaking  
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Battleships and New Scout Cruisers

# Chance Vought Corporation

Long Island City, New York

OCTOBER 1, 1923

# AVIATION

VOL. XV. NO. 14

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# AVIATION

Vol. XV

OCTOBER 1, 1933

No. 14

### St. Louis as an Airport

**L**AST year, at the Detroit races, it was the unanimous feeling that the home of the automobile had given no reason to the maximum in aeronautical interest. As might have been expected, St. Louis has exceeded the greatest hopes of its citizens, and the public will see a spectacle of aircraft performance that will be unparalleled in the history of aviation. At Detroit, the lighter-than-air development played no part, but at St. Louis, weather permitting, the ZRI will make her closing trip in the Mississippi Valley. There is no scenario for going farther into details. St. Louis has elucidated aeronautical progress in this country in a way that will ever remain to her credit.

The future position of St. Louis in the aeronautical activities of the country has been made secure by the creation of a permanent airframe. Strategically located, it will always be regarded as an aerial terminus of the first class. The famous and public-spirited men, who have made this permanent movement in the future of aviation, will some day be placed in the same group that second earth opportunities to make their cities racing centers.

In the farthest result of the great effort will be the national interest that will be created. Prominent men representing finance and industry will interpret aviation in the terms of their special needs. Government officials will lay before the country the opportunities for aviation to be of service to the various Departments. The Air Institute will give not only this broad consideration to aeronautics, but the technical and commercial phases of air development will also be given expression.

And the National Aeronautic Association will mark the end of the first year of existence. The deliberations of this august gathering will have a profound effect on the future of aviation. No man must be made that will shape the policy of the organizations and, thereon, in no small way, will effect the growth of aeronautical interest all over the country. We looked at these bold birds the local and national viewpoint, the next days of October will focus all eyes on St. Louis. Aviation expresses the congratulatory word at this time and will give in forthcoming times a complete and accurate account of all the happenings of the meet.

### Service Ships Versus Racers

**A**FTER the St. Louis meet it will be advisable for the Army and the Navy to take stock of the aeronautical conditions that the last two years has brought out and reach a decision regarding the utilization of the speeds attained. Being aircraft are an excellent laboratory development and have brought to our surface designers and constructors the time of being ahead of the world in aeronautical skill.

A year has elapsed since the records were made in Detroit and except for a few experimental types, no translation of the racing types into service types has been made. Unless this is done the expenditure for the races will not be utilized to its maximum.

It should be made clear to the public that these races are not service machines and that the records made do not mean that this speed could be made use of in warfare. Unless this is done, a false feeling of security will be created with the result that when appropriations are needed for service aircraft, they will be refused on the grounds that we have the fastest machines now, so why worry.

The competitive spirit that has been developed between the Army and Navy as a result of these races is a real sign of progress. The Army, last year, freely spent money to develop its record-breaking airplanes. This year the Navy will have the new machines. During accidents, the Navy will probably pass from the Army to the Navy, for the designers of last year's ships have put into the 1933 season the results of much racing experience.

Another fact that should not be lost sight of is that this year the race will be in reality a contest between the Wright and Curtiss designers. The leading manufacturers have their two hundred names and will have many members of the early struggles for supremacy in the air.

There is only one feature of the various events that may cause disappointment and that is the almost total lack of new service planes. The last twelve months have produced little new along this line except in novel types of airplanes and while the experimental aircraft that will be seen will probably be of great interest, it is only reasonable to hope that at earlier meet the new service equipment will be put in competition.

### Airship Operation

**A**DRIEL MOFFAT recently said that the airship ZRI "has a mission in peace which will make her a factor in the commercial expansion of our country. As a means of transportation for passengers and freight, as a means for transoceanic travel and for purpose of exploration, geological survey, and forest patrol, the rigid airship stands in a class alone, and is destined to have a wide sphere of usefulness." He also states that the data that is being kept will be available as a basis of study for successful airship development. No more important information could be made available as a guide to the public. The airship has had very optimistic friends and their opinions has caused a great deal of money to be invested in this branch of aviation. Official data furnished by the Navy Department as to cost of equipment, operation and depreciation will give just the same criterion as the data furnished by the Air Mail Service.



## ENTRIES FOR FLYING CLUB OF ST. LOUIS TROPHY

Pilot	Plane	No.	Engine	Hp.	Entrant
Robert F. Hewitt	Farmen "Sport"	1	Anzani	60	Ludington E. Co.
Charles E. Jones	Curtiss Oriole	2	Curtiss OXS	60	Curtiss E. Co.
Lewis H. Sperry	Stearman	3	Lewis H. Sperry	60	Lewis H. Sperry
Edmund T. Allen	D A 4 plane	27	La Rhone	60	Henry A. Berthier
Wm. D. Robertson	Special plane	38	Curtiss OXS	60	Robertson Aircraft Corp.
Frank H. Robertson					
Walter K. Lee					
Perry O. Hutton					
John K. La Grosse	Hawker PUL	28	Curtiss OXS	60	Johann A.P. & Eng. Co.
	Laerd Swallow	35	Curtiss OXS	60	Stalling Oil & Ref. Co.
	Rogers-Day	66	Curtiss OXS	60	John K. La Grosse

acrometers used. No claim may be made against your fuel, loss of time, etc.

The contest is open to any make or type of aircraft. The flight may start at any time after Sept. 20, 1933, but must be completed on or before midnight, Sunday, Sept. 30, 1933.

Prizes will be awarded on the basis of number of points received by contestants. First prize will be \$500.00 in cash and a silver trophy cup. Second prize will be \$250.00 in cash. Third prize will be \$125.00 in cash. Fourth prize will be \$75.00 in cash. In case of a tie the winner will be determined by lot. The decision of the Contest Committee of the Flying Club of St. Louis is final.

## Flying Club of St. Louis Trophy

Monday, Oct. 1, 1933

This Trophy, donated by the Flying Club of St. Louis, shall be competed for by airplanes of 90 hp. or less, and shall become the permanent possession of the entrant of the winning airplane.

## CONDITIONS OF CONTEST:

(a) In order to eliminate interest among civilian pilots, Army and Navy pilots will not compete in this race.

(b) Engines must be 90 hp. or less. (Horsepower as given by the Manufacturers' Data Book of the U. S. Air Service as McCook Field will be taken as standard.)

(c) All airplanes must carry a total load of 240 lb., evenly distributed as follows: this load to consist of pilot and one passenger, with sufficient ballast, if necessary, to bring their weight up to the required amount. It is required by the Contest Committee that all contestants carry ballast in lieu of passenger, in order to eliminate hazard in this respect. Under no circumstances may open cockpit be covered.

(d) General condition and design of plane to be such that in the opinion of the Contest Committee it is safe and not a menace to other contestants or spectators. The Contest Committee reserves the right to refuse to admit any airplane which does not comply with these requirements.

(e) At the time of entry the entrant must supply the Contest Committee with a statement, giving the make and stroke of the engine to be used in the race, this statement to be properly countersigned by a notary public. The Contest Committee reserves the right to check the measurements of any engine.

## ENTRIES FOR LIBERTY ENGINE BUILDERS' TROPHY

Pilot	Plane	No.	Motor	Hp.	Entrant
Ray D. C. Allen, U.S.N.	Curtiss BT	3	Curtiss CD12	400	Thomas of Aeromarine
Ed La. D. H. Black, U.S.M.C.	Vought V391	4	Liberty	400	Thomas of Aeromarine
La. S. H. Connelley, A.S.	Leaper	30	Liberty	400	U. S. Army Air Service
La. Haines N. Brown, A.S.	CG4	31	Liberty	400	U. S. Army Air Service
La. C. McMahon, A.S.	CG4	32	Liberty	400	U. S. Army Air Service
La. L. H. South, A.S.	CG5	33	Liberty	400	U. S. Army Air Service
La. V. J. Mingo, A.S.	NDA1A	34	Wright	380	U. S. Army Air Service
La. W. T. Laven, A.S.	NDA1A	35	Wright	380	U. S. Army Air Service
La. H. K. Roney, A.S.	DH4C	36	Liberty	400	U. S. Army Air Service
La. M. M. Custer, U.S.C.	DH4C	37	Liberty	400	U. S. Army Air Service
La. W. H. Haskins, A.S.	DH4C	38	Liberty	400	U. S. Army Air Service
La. J. J. O'Connor, A.S.	DH4C	39	Liberty	400	U. S. Army Air Service
Mr. Ray S. Brown, A.S.	Laerd	40	Liberty	400	U. S. Army Air Service
Capt. Scott Coffey, A.S.	Boeing-Fokker DT*	41	Boeing F13 1937	400	U. S. Army Air Service
La. W. H. Brookley, A.S.	Curtiss*	62	Curtiss D12	400	U. S. Army Air Service

\*Two-seater observation plane.

selected and to demand the removal of a cylinder for this purpose. The decision of the Contest Committee will be final.

## DISTANCE:

100 kilometers (62.137 miles), three times around a closed course of 33.385 kilometers (20.747 miles), starting at St. Louis Field, thence north to the first turning pylon, thence west to the second turning pylon, and return to St. Louis Field.

## NUMBER OF CONTESTANTS:

Maximum number: Four.

Maximum number: Sixty.

## CASH PRIZES:

First prize \$500, second prize \$300, third prize \$250.

## Liberty Engine Builders' Trophy

Monday, Oct. 1, 1933

This Trophy, donated by the builders of the Liberty engine, shall be competed for annually by observation type (two-place) airplanes or flying boats.

## CONDITIONS OF CONTEST:

(a) Factor of safety: 8 at loaded for start of race.

(b) Air speed greater than 90 m.p.h.

(c) Carrying the following "contest load" and, in addition, over 340 lb. "Contest loads" shall be determined from the following formula:  $\text{cubic inch displacement} \times .0001$  for the 1933 hp. Liberty engine.

$\text{---} \times \text{cubic inch displacement of motor used} \times .0001 = \text{Contest Load.}$

Note: The reciprocating and parabolic may be included in the "contest load", however, fuel, accessories, gas, oil and water for the engine are excluded. Such tanks, gas and accessories may be mounted. At the time of entry the entrant must supply the Contest Committee with a statement giving the make and stroke of the engine to be used in the race, this statement to be properly countersigned by a notary public. The Contest Committee reserves the right to check the measurements of any engine entered and to demand the removal of a cylinder for this purpose. In case there is a question of the cubic inch displacement of the engine, the Contest Committee



Farmen "Sport" (50-60 hp. Anzani), entered in the Flying Club of St. Louis race



Bellanca CF cubic plane (50 hp. Anzani) entered in the American Country Club of Detroit race



U. S. Army-Fokker T2 (400 hp. Liberty), entered in the Merchants Exchange Trophy race







# Army and Navy Entries in the Pulitzer Trophy Race

Oct. 3, 1923

AVIATION

401

## Thin Wing Biplanes in Great Majority—Most Heavily Powered Racers in the World to Compete

This year's Pulitzer Trophy Race, which will be held Oct. 7 at St. Louis, will be particularly interesting on several counts. To begin with, the two Navy-Wright Fighters, built by the Wright Aeronautical Corp. for entry, by the Bureau of Aeronautics, Navy Department, in the Pulitzer race are with their 750 hp. Wright T3 engines the highest powered airplanes in the world, with the possible exception of the Fiat Basso, about which little definite information is available.

streamlining on one hand and from added horsepower on the other.

Following are the principal characteristics of the Army and Navy entries in the Pulitzer Trophy Race:

### The Navy-Wright Fighters

The Navy-Wright Fighter was developed by the Wright Aeronautical Corp. of Paterson, N. J. Its tests at Mitchell



Navy-Wright Fighter powered with a 750 hp. Wright model T3 engine, one of the four Navy entries in the Pulitzer Race. In tests at Mitchell Field this ship covered a 1-kilometer course at an air speed of 238 mi./hr.

In their speed tests at Mitchell Field, these ships gave a very suggestive demonstration of both speed and maneuverability. These several planes may, therefore, be expected to be strong contenders in the Pulitzer race.

The two Army-Curtis BSC1 Racers, which with the two Navy-Wright Fighters complete the Navy team, in the Pulitzer race, are likewise equipped with a remarkably effective motor. Fitted with the new Curtiss racing engine, the BSC1 Special, which develops 500 hp. at 2300 r.p.m., these ships, despite their lower horsepower, will be very dangerous competitors for the Navy-Wright Fighters.

The Army Air Service will be represented in the Pulitzer Trophy Race by three entries, two of which will be last year's Army-Curtis Racers, while the third will be a Verville-Sperry racer. Two of these ships, one of the Army-Curtis Racers and the Verville-Sperry, have been equipped with the new Curtiss D12 Special engine, while the other Army-Curtis racer will have last year's 465 hp. Curtiss D12 engine. It will be interesting to compare the performance of the Army and Navy Curtiss Racers, for they will afford valuable assistance in to how more speed can be obtained from better

field indicated clearly that it will be one of the foremost contenders for the honor of winning the Pulitzer Trophy Race.

It is a single-engine tractor biplane, equipped with a Wright box compression Model "E3" engine of 700 hp. The overall length is 22 ft. 4 in.; the span 22 ft. 6 in., and the overall height 7 ft. 11 in. The ship is 4 ft. 4 in., and there is no stream dihedral or sweepback.

The wings are set at an angle of incidence of 9 deg. and are built up of spruce beams covered with two-ply spruce planking. There are two lower wing panels and a single upper panel. The outboard struts are of I form and built up of spruce, while the center strut is a single vertical steel framework, constructed of four welded steel tubes connected with fitting. The ends of the streamline housing are carried by suspension steel fittings, which permit the wires to pass through the wings, so that their ends are readily accessible for adjustment sideways entirely out of the stream.

The horizontal stabilizer and vertical fin are built up of spruce ribs, with mahogany ply covering, while the elevator and rudder have steel frames with fabric covering. The horizontal stabilizer is adjustable on the ground.

The fuselage is of the monocoque type, with a two-ply spruce shell and four spruce longerons, together with ply-



Side view of the Navy-Wright Fighter entered in the Pulitzer Trophy Race.

wood bulkheads sufficiently close to ensure that the shell will retain its shape under the most severe conditions.

The engine mounting is of the same construction below the main fuselage as the fuselage, above the bottom there is a web diagonal steel tubular strut on each side running down to the top of the fin wall.

The landing gear is of steel with wood fairing of the large type, with two tubular V struts, two steel spooler tubes, two main axles and elastic cord shock absorbers.

The propeller has three blades of forged duralumin, fitted to a steel hub in such a way that the pitch of the blades can be readily altered to secure the greatest efficiency. The blades are of such diameter action that those of the conventional wood propeller, and on account of the unusually curved construction, the propeller is remarkably well fitted to withstand the very severe loads which come in during the race.

The cooling water flows from the engine into four radiators

of the wing surface type, on each side of the upper wing panel and one on each side of the lower wing panel. The water first enters the radiator on the leading edge of the wing and thence flows into cooling elements on the upper and lower surfaces of the wing. Each element is constructed of two sheets of thin brass riveted together, the lower sheet slightly corrugated and the upper very deeply corrugated. The secret of this construction, the water flows through the elements in very thin streams, so that the cooling power of the radiator is very high, while at the same time the effect of the induction in the aerodynamic characteristics of the wing is entirely negligible. The construction of the radiators is such that it is a comparatively simple operation to remove any damaged cooling element from the radiator for replacement without disturbing the wing or the other elements of the radiator. A baffle in the leading edge of the wing collects the cold water from the cooling elements and from here it flows to the reserve water tank in the center of the upper wing panel. The reserve water pump draws directly from the reserve water tank.

As might be expected, the entire machine presents an unusually clean and trim appearance. Although this fact makes the design a somewhat grossly fast racing, the general characteristics are such that the ship can, without much difficulty, be adapted to a fighting type of superior performance.

A good distinguishing mark of this ship is eight are the prominent streamers containing streamers with the 1 type navy/air stream which one of the Curtiss racers later, and the following navy color scheme adopted by the Navy.

### The Army Curtiss Racers

In 1923 Army-Curtis Racers are identical with those entered in last year's Pulitzer race, save that one of them is equipped with the new Curtiss D12 Special 500 hp. motor.

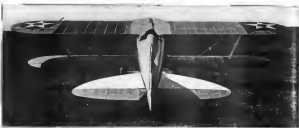
The body structure is covered with "Curtiss-ply," a strong light body material of laminated spruce 3/32 in. thick. The lanes of the fuselage were designed around the smallest possible cross-sectional area, merely large enough to clear the engine and to accommodate the pilot.

The weight of the body complete is only 127 lb. due to the employment of duralumin fittings, together with the use of the Curtiss-ply construction which proved so successful in previous use.

Light weight, rigidity and extreme stiffness are features of the multi-spar Curtiss-ply covered wings. The Curtiss 27



Army Curtiss racer, fitted with a Curtiss D12 Special engine, developing 500 hp.—one of the Navy's few entries in the Pulitzer Race. Fitted by Lieut. Harold J. Brown (above) this ship covered a 1-kilometer course at an air speed of 244 mi./hr. on Sept. 17.



New view of the Navy Curtiss racer entered in the Pulitzer Trophy race

wing section is now developed by engineers from data obtained from testing high speed curves on the definite wind tunnel standard at Dayton City.

The wing radiator, designed and perfected by the Curtiss Co., deserves special mention. Many tests were made with them corrugated sheet brass radiator sections which replace the surface of the wings. With this system, advantage is taken of the flow of air around the wings, which cool the engine water without any extra head resistance whatever. They were successfully used on the Curtiss-Pulitzer Racers last year.

The fixed radiator, in and stabilizer are of the Curtiss-type radiator construction, with an external heating. Control sections, address, fuselage and rudder, are of steel and duralumin. All controls are actuated by means of internal cables and operating tubes.

The single sheet type of landing gear used has the strength and rigidity of the conventional "V" chassis, yet the construction has been reduced two-thirds. Even the wheels and hub caps have been streamlined. Wheels are provided with stepped discs of aluminum which covers the spokes and hubs, being moving as applied over the wheel and tire in such a manner that a smooth and uniform is presented to the air stream.

The tail skid is made of laminated strand growth balsa, forming an effective shock-absorbing tail adding little to the total resistance of the machine.

#### CHARACTERISTICS

Engine, 350 h. p.	Wing span, 35 ft.
Length, 17 ft. 11 in.	Height, 10 ft. 6 in.
Wing area, 110 sq. ft.	Wing loading, 32 lb. per sq. ft.
Engine, 350 h. p.	Wing loading, 32 lb. per sq. ft.
Engine, 350 h. p.	Wing loading, 32 lb. per sq. ft.
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Engine, 350 h. p.	Wing loading, 32 lb. per sq. ft.
Engine, 350 h. p.	Wing loading, 32 lb. per sq. ft.

#### The Navy-Curtiss Racers

Constructively the Navy-Curtiss racers differ very little from the Army-Curtiss racers, the principal innovations being a somewhat rounder and better streamlined body, the placing of the wing close to the top of the fuselage and a simplified landing gear. The latter carries the shock absorbers

directly in the wheels, whereby a lot of parasite resistance is eliminated.

Another novel feature in the Navy Racers is that the side view of the wings are attached in the leading gear instead of to the fuselage as in the case with the Army racers.

The power plant consists of the new 500 hp. Curtiss D12 Special engine with a fixed down-draft propeller and the wing radiator of last year.

Although the two types of Curtiss racers look very much alike, nevertheless, in flight, it will be fairly easy to distinguish them from one another, for the Navy ships have golden wings and royal blue body with a "tailor" propeller, while the Army ships have a black fuselage with golden wings and golden propeller.

#### The Army Variable-Sperry Racer

The third of the Army Air Service entries for the Pulitzer Race is a redesigned Variable-Sperry, B3 Racer, one of which type participated in the Pulitzer Race held at Selfridge Field, Mich. in 1931. The first of the B3 racers made an average speed of 181 mi/hr. in the last Pulitzer Race. On a later date at McCook Field, Dayton, Ohio one of the B3 Racers made a speed in level flight of 191 mi/hr.

One of these B3 racers was redesigned by the Air Service Engineering Division, altered Variable-Sperry type, and rebuilt by the Lawrence Sperry Aircraft Co. at Farmdale, L. I. The major changes on this racer are: Installing a Curtiss D12 Special, 500 hp. engine and wing radiator which replaced the Wright 300 hp. engine and two Lambda model H radiators, and changing the wing type to an elliptical form which slightly increased the wing area.

In a level speed test held at Mather Field, Sept. 2, the redesigned B3 racer made a speed about 54 per cent higher than the original B3. The factors contributing to the higher speed are more power and less parasite resistance. The Curtiss D12 engine with the large bore (16 in.) cylinders and high compression (25.5 compression ratio) delivers about 48 per cent more power than the engine previously used. The parasite resistance was reduced by reducing the frontal area of the fuselage which was made possible by the incorporation of the Curtiss D12 engine. The greatest reduction in parasite area was gained by using Curtiss type wing radiators which replaced the Lambda type. This reduced the flat plate area of the whole machine about 28 per cent.

The Curtiss type wing radiator used on the redesigned B3 is built up in four sections: the upper and lower surface of each side making an independent unit. The upper and lower

#### CHARACTERISTICS OF ENTRIES FOR PULITZER TROPHY

Entry	Engine	Wing Span	Wing Area	Wing Loading	Wing Area	Wing Loading
Navy-Weight T3	Weight T3	700 22'0" 21'4"	sq. ft. 1000 10	sq. ft. 1000 10	sq. ft. 1000 10	sq. ft. 1000 10
Navy-Curtiss D12C1	Curtiss D12 Special	500 12'0" 19'10"	145	sq. ft. 200 10	sq. ft. 200 10	sq. ft. 200 10
Army-Curtiss B3	Curtiss D12 Special*	500 19'0" 18'10"	133	sq. ft. 200 10	sq. ft. 200 10	sq. ft. 200 10
Army-Variable-Sperry Racer	Curtiss D12 Special*	500 18'0" 22'0"	145	sq. ft. 200 10	sq. ft. 200 10	sq. ft. 200 10

\*One of these racers is equipped with the Curtiss D12, 650hp. engine, the same as used in last year's race. The characteristics of this plane are not available.

sections are securely held together at the leading ends by  $\frac{1}{2}$  in. brass pins. The panel sections of the radiator are supported on the wing surface by 1/16 in. steel wires placed in rows about 8 in. apart. These wires are threaded over the surface of the radiator and through strap metal lugs which are secured to the wing covering and extend through cylinders welded in the radiator panels. To support this type wing radiator, it was necessary to cover the entire surface of the wing and 2/32 in. plywood, which was then covered by the radiator. This method increased the weight of the wing slightly and it also increased the stiffness of the wing which is necessary and is difficult to get with an internally braced wing. The wing stiffness is required to eliminate surface area deformation caused by the center of pressure travel.

The engine oil is not cooled by the nitrogen, but, by the water. As the water leaves the radiator, it passes through the oil temperature regulator or cooler. This system permits of maintaining a more constant oil temperature in all conditions of flight. The oil temperature regulator is built up around a 5 in. cartridge core. The water passing through the tube and the oil around the tubes a by-pass around the temperature regulator is provided so the oil may be by-passed when cold and stiff to eliminate excessive pressure in the engine oillet oil line.

The external appearance of the plane was improved and better streamlining of the nose was obtained by using an

aluminum spinner on the propeller. Also, by the installation of the Curtiss D12 engine, it was possible to lower the thrust line 3 in. below the position in the original B3. This change brought the thrust line close to the center of resistance and gave a more symmetrical outline to the upper and lower surface of the fuselage.

All plane sections are solid connected the same as on the original B3.

There was an change made in the chassis which folds into the lower surface of the wing and is operated from the cockpit by a hand crank on a shaft and gear system. The position of the chassis is regulated by a mechanism in the cockpit.

The tail section and fuselage is the same as the one in the original B3. As this fuselage was originally designed to take an engine with larger overall dimensions than the Curtiss D12, it would be possible to build a speedier plane with the same wing, chassis, tail and gear on a narrower fuselage which would further reduce parasite resistance. This fact should be considered when comparing the speed of this racer with the plane using the same power plant and designed for this year's race.

The Variable-Sperry racer will be the only monoplane of machine type on the Pulitzer race unless the Fiat racer, which is believed to be a biplane, turns out to be a monoplane, too.



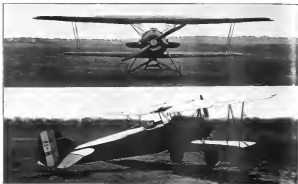
First test of the modified Variable-Sperry Racer (300 hp. Curtiss D12 Special engine) entered by the Army Air Service in the Pulitzer Race





# Huff-Daland Model TA6 Advanced Training Plane

Safety, Simplicity and Ease of Alignment and Assembly  
Chief Features of New Plane



Two views of the Huff-Daland TA6 advanced training plane equipped with a 200 hp Wright-Lawrence J1 radial engine

Huff, Daland & Co., Inc. plane is to be represented by two airplanes at the National Air Races in St. Louis. One of these is the TA6 advanced training plane, which was purchased and tested by the U. S. Army Air Service in the early Spring of 1925, to be piloted by Lt. H. R. Harvey; the other is the new "Pittet", Model 4, the company's latest contribution to the commercial field, piloted by George B. Post. The two planes will fly to St. Louis from Oshkosh, W. V., where the TA6 has been undergoing studies and tests in preparation for the production of five TA6 training planes on a recent order received from the Army Air Service.

Both the TA6 and the "Pittet", Model 4, are monoplanes carrying the same type of wings, fuselage, and landing gear as the TA6, but are equipped with Wright E or Curtiss OX5 engines, instead of the Wright Lawrence J1, with which the TA6 is powered. Details and specifications of these airplanes are not yet completely prepared for publication, but it is felt that the following description of the TA6, which appears in general to all three machines will prove of timely interest.

Designed as an advanced plane for the U. S. Army where rugged construction and protection for student and instructor is one of serious considerations, the Huff-Daland TA6 is a standard fuselage, welded into a complete and without the use of inter-lay wires was chosen as the most reasonable type. Alignment problems during manufacture presented unusual difficulty in construction, but the fact that this type of fuselage

completely eliminated the possibility to give and take, and the necessary fit alignment due to hard usage and climatic changes in service, more than compensated for the difficulty involved.

The wings are of the pure cantilever type, tapering in plan as well as in section, based upon the USASB aerfoil, and resulting in a very stiff structure with excellent characteristics for maneuverability and ease of lateral control. Both wings are built upon continuous box spars, fixed with double diagonal brace planking on either side. The upper wing is supported by two trapezoids and two adjustable flying struts of steel tubing, while the lower wing is raised into a rams in the fuselage, and driven home by two tie rods bolted in place. Ailerons are carried in the upper wing only and built upon torque tubes operated by push and pull rods from the control system in the fuselage. Elevator and stabilizer are of unusually high speed ratio resulting in a large amount of inherent stability without any sacrifice of definite control, and are moved by wires in conventional manner.

The landing gear is a noteworthy example of the tripod type which does away with the conventional axle between the wheels, and the consequent danger of shaking and set of one in tail grain at grass. Shock absorbers are taken care of by two special axle bolsters to the axle which are readily accessible and can be easily removed for repair. Interchangeable pneumatic and rubber axles have both been used with complete success.

October 3, 1925

AVIATION

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The fuel system consists of two 22 gal. aluminum crash-proof gas tanks located between the spars of the upper wing on either side of the fuselage. Gravity tanks are carried from front and rear of both tanks through a three-way valve and connected to service streamer to the carburetor, resulting in remarkably simple and fool proof design.

The TA6 was first flown by Art Smith from the front section of the St. Lawrence in February 1925 in acceptance trials for the Army Air Service. Since then it has been in constant service at Oshkosh and Dayton, receiving excellent reports on its maneuverability and general flying qualities and from a maintenance stand point.

The following specifications and data are taken from the official performance of the TA6, as made at McCook Field under fully loaded conditions. Subsequent tests with a more powerful propeller have materially increased both climb and speed.

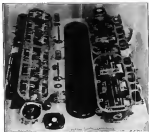
## CHARACTERISTICS OF THE HUFF-DALAND TA6 TRAINING PLANE

Engine, Wright Lawrence J1, 200 hp.  
Wing, upper wing 33 ft.  
Wing, lower wing 27 ft.  
Landing gear, 11 ft. x 11 ft.  
Wing, tip chord 10 ft.  
Wing, root chord 14 ft.  
Wing, tip chord 10 ft.  
Wing, root chord 14 ft.  
Wing, tip chord 10 ft.  
Wing, root chord 14 ft.  
Wing, tip chord 10 ft.  
Wing, root chord 14 ft.

## A New Powerplant

Wright Model T3 Aviation Engine

The Wright Model T3 engine represents the latest development in the 800 hp. water-cooled class of aviation engines, and is a further development of the Wright T2 engine which has been in production over a year. It is a 90 deg. Vee engine, with cylinders 5 1/2 in. x 5 1/2 in., which gives a displacement of 1047 cu. in. The engine has two power ratings, 515 hp. at 1800 rpm. and 800 hp. at 2000 rpm. The weight dry, with complete equipment, is 1100 lbs., which gives a weight ratio in the latest case of 17.7 lb./hp. This is a 10 per cent reduction in the weight per horsepower of the Liberty engine. The Model T3 was brought out in response to the demand of the Navy Department for a power plant of considerably greater power than the Liberty, but of about the same size and weight. Not only was a larger power unit required, but it was felt



Cordell bearing assembly of the Wright T3 engine

that it was now possible, after approximately six years of development work, to build a more efficient and desirable aircraft engine.

The T3 differs from the T2 in detail only, the bore and stroke of the two engines being identical. Greater power has been secured by an improved intake system. Smaller overall dimensions have been secured by a careful redesign, so that the engine is 5 in. shorter than the Liberty. Greater durability, even though the T3 is provided by certain changes in material and by increasing the size of certain bearings. Maximum smoothness in operation has been secured by a 3/4 in. crankshaft, a very rapid crankshaft and cylinder construction, as well as improved valvetrains.

The new engine is suitable for use in any type of plane requiring its power. Due to the low weight per horsepower and small overall dimensions, coupled with the ability to run for long periods without overhaul, it is particularly adaptable to bombers, seaplanes, torpedo planes and large ships due to the small service, as well as low-water flight.

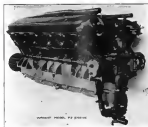
## Real Commercial Aviation

First Flights Carried by Air

On Friday, Sept. 24, at midnight W. L. Smith took off in a Curtiss machine from Van Courtlandt Park, carrying photographs of the hospital and other pictures of the forward ward. He served Cleveland and Chicago and arrived at Omaha at 5:45 p. m. Saturday. The pictures and notes were delivered to a representative of the News Enterprise Association at Omaha, who took them aboard a special train on the Great Pacific for Chicago, where Kenneth Gager piloting a Red Curtiss triplane received them. He left Chicago at day-break Sunday morning, stopping at Salt Lake City and arriving at Fresno at 5:30 Sunday afternoon, making the total delivery time across the continent forty-one and a half hours.

This quick service made it possible for the Newspaper Enterprise Association to send its papers far ahead of all competitors. They registered a record of five hours at Cleveland, Chicago and Omaha, and a much longer time at the Salt Lake.

Curtiss is withdrawing its cross-country flying operations the same expediency that is recognized in the industry on account of its engineering and manufacturing ability.



Wright Model T3, 700 hp. engine















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NAVY CURTISS RACER

In 1909 Glenn H. Curtiss won the Gordon Bennett Race, the French speed classic.

Since that date the Curtiss organization has led the world in the design and construction of aeroplanes and motors.

In 1921 the Navy Curtiss Racer with a Curtiss motor won the Pulitzer Race at Omaha and established the world's speed record.

In 1922 in the Pulitzer Race at Detroit the Army Curtiss Racers with Curtiss motors took first and second places, and the Navy Curtiss planes (of 1921) took third and fourth places, again establishing world's speed records for the various distances covered.

In 1923 the Navy Curtiss Racers with Curtiss motors are still faster.



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